



## Effect of chemical composition on the semisolid tensile properties and hot tearing susceptibility of AA6111 DC cast alloys

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### ABSTRACT

The semisolid tensile properties of two AA6111 direct-chill cast alloys (A and B) have been studied. The Cu, Mn, and Si contents of alloy A are higher than those of alloy B. The microstructures of the alloys were analyzed before tensile testing and after tensile fracture. Isothermal holding was performed in the temperatures of 510, 520, 535, 552, 564 and 580 °C for 1 h to study porosity/void formation in both alloys. Tensile tests were conducted near the solidus temperature in the temperature range of 450–580 °C at a strain rate of  $10^{-4} \text{ s}^{-1}$ . The strain during tensile testing was measured using the digital image correlation method to obtain reliable stress–strain curves. The results revealed that the tensile strengths of the alloys gradually decreased to zero with increasing temperature to arrive at the zero-stress temperature, whereas the strains at the failure decreased sharply with increasing temperature until zero-ductility temperature (ZDT) was reached. Moreover, the failure strain of alloy B at any given testing temperature was higher than that of alloy A. Non-mechanical and mechanical hot-tearing criteria were used to study the hot-tearing susceptibilities (HTSs) of the alloys. Considering the mechanical criterion, the ZDT and brittle temperature range of alloy A were lower and larger than those of alloy B, respectively, indicating that the HTS index of alloy A was higher than that of alloy B.

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## 43 Introduction

44

45 Hot tearing, a common cast defect that occurs during  
46 the last stage of solidification, involves a continuous  
47 solid network of dendritic grains surrounded by a  
48 liquid film and pockets [1]. Mush structures are often  
49 subjected to tensile stress due to the thermal gradient  
50 and solidification shrinkage during casting. When the  
51 strength of the mush structure is insufficient to sus-  
52 tain the applied thermal stress, cast defects, such as  
53 hot tearing and porosity, occur [1–3]. Hot-tearing  
54 susceptibility (HTS) depends on several factors, such  
55 as solidification interval, microstructure develop-  
56 ment, eutectic feeding ability, and mechanical  
57 response of solidified microstructure [1–3].

58 The partial remelting method of as-cast samples  
59 during tensile testing has been widely adopted to  
60 investigate the mechanical response of semisolid Al  
61 alloys because it induces similar stress–strain condi-  
62 tions to those during solidification and provides  
63 quantitative stress and strain results for semisolid  
64 alloys [4, 5]. However, partial remelting of samples  
65 subjected to tensile tests presents some challenges,  
66 such as high thermal gradients and strain localiza-  
67 tions, even at small strains [6]. Several researchers  
68 have investigated the semisolid tensile properties of  
69 various Al alloys, including AA5182 [7–9], AA3014,  
70 AA6111 [10], and AA6061 [11]. Previous studies have  
71 indicated that tensile tests should be performed in the  
72 strain-rates range of  $10^{-5}$ – $10^{-3}$  s<sup>-1</sup> to simulate the  
73 direct-chill (DC) casting process [7, 12, 13]. Hot-tear-  
74 ing studies require semisolid tensile tests at low liq-  
75 uid fractions ( $f_L < 10\%$ ), such that the specimens  
76 retain their original shapes as solid [11]. A short  
77 holding time at a semisolid temperature (within a  
78 few minutes) and high heating rates are recom-  
79 mended for partial remelting tests to minimize the  
80 effect of back-diffusion [4].

81 HTS is strongly related to the alloy composition,  
82 and the addition of small amounts of alloying ele-  
83 ments can affect the HTS indices of Al alloys [1, 2]. Fe  
84 is usually considered a harmful element in Al–Mg–Si  
85 alloys because coarse and large Fe-rich intermetallics  
86 can hinder metal feeding during the last stage of  
87 solidification [14]. It is reported that the hot tearing  
88 susceptibility of Al–Mg–Si–Fe alloy reaches to its  
89 maximum at 0.2% Fe [15]. By adding Mn to Al–Mg–Si  
90 alloys, the Chinese-script Fe-bearing intermetallics  
91 were formed to suppress the formation of coarse  $\beta$ -Fe

intermetallics, which facilitated the formation of solid  
bridges and the flow of liquid metal within semisolid  
structure [16]. It is also reported that increasing Si  
content enhanced hot tearing susceptibility, which  
reached to its maximum at 1% Si [17]. On the other  
hand, adding Cu to Al–Mg–Si alloys improves the  
mechanical strength of cast parts and causes the  
formation of Cu-bearing intermetallics (e.g., Al<sub>2</sub>Cu  
phases) [18]. Cu-bearing intermetallics can signifi-  
cantly decrease the melting point and increase the  
solidification interval of Al alloys [18, 19]. HTS is also  
related to the amount of low-melting-point eutectic  
liquid in the later stages of solidification. Hot tear can  
be initiated in solid dendritic networks once the  
volume fraction of the liquid phase is in the range of  
2–5% [20]. The relationship between the hot-tearing  
resistance and eutectic content depends on the dis-  
tribution of the eutectic liquid to the grains [1, 21].  
The formation of eutectic liquid along the grain  
boundaries renders the grains brittle and promotes  
the propagation of hot tears [22, 23].

The criteria used for predicting the HTS of Al  
alloys can be classified as non-mechanical and  
mechanical models [21, 24]. The non-mechanical cri-  
teria account for the fluid flow and healing of the  
structure depending on the feeding conditions. For  
example, Kou [25] proposed a non-mechanical model  
with a crack sensitivity index based on the steepness  
( $dT/df_s^{1/2}$ ) at  $f_s^{1/2} \approx 1$ , used to evaluate the relationship  
between the location of the peaks of the crack sensi-  
tivity curves and alloy chemistry. Conversely, the  
mechanical criteria—including stress-, strain-, and  
strain-rate-based models—emphasize the importance  
of strengths and strains developed during the inter-  
dendritic separation and bridging stages of solidifi-  
cation. Several hot-tearing models consider a critical  
temperature range, where the possibility of hot tear-  
ing increases [15, 21, 26, 27]. This temperature range,  
known as “brittle temperature range” (BTR), spans  
from the zero-ductility temperature (ZDT) to the  
zero-strength temperature (ZST); in this range, the  
material can sustain its strength without further  
straining [10, 28]. The wider the BTR, the larger the  
HTS index. The ZDTs of the alloys are typically lower  
than their ZSTs. The presence of minor elements,  
such as Cu, can significantly affect the BTR because  
these elements can induce a series of complex eutectic  
reactions toward the end of solidification [15, 29, 30].

Al–Mg–Si AA6111 wrought alloys are widely used in the transportation industry owing to their high strength/weight ratio, good corrosion resistance, and reasonable formability. AA6111 wrought alloys are primarily manufactured through ingot metallurgy via DC casting, followed by thermomechanical processes, such as rolling and extrusion. The primary alloying elements in the AA6111 alloys are Mg, Si, Cu, and Mn. Considering the multicomponent nature of the alloying elements, AA6111 alloys exhibit a wide solidification interval and generate numerous as-cast microstructures with different intermetallic phases during solidification; therefore, they are susceptible to hot tearing and porosity during DC casting. These cast defects are harmful and limit DC casting productivity.

In this study, the effect of the chemical composition of AA6111 DC cast alloys on their tensile response above the solidus temperature was studied. Considering the high sensitivity of the tensile samples to the high test temperature in the semisolid region, the strain was measured using the digital image correlation (DIC) method to ensure accurate results. HTS was further investigated using two primary criteria: a non-mechanical criterion, such as that developed by Kou [25, 31], and a mechanical criterion based on the BTR [10, 32–34].

## 167 Experimental

### 168 Materials

Two AA6111 alloys with different chemical compositions (alloys A and B) were selected, owing to their large solidification ranges ( $\sim 142$  °C based on Scheil calculations) and high HTS [10]. DC cast alloy ingots (590 mm  $\times$  185 mm  $\times$  70 mm) were provided by the Arvida Research and Development Center of Rio Tinto (Saguenay, Quebec). The chemical compositions of the alloys were determined by optical emission spectroscopy; the results are summarized in Table 1. The metallographic and tensile test samples

were cut from the mid-center areas of the DC cast ingot parallel to the casting direction, as shown in Fig. 1, which well represented the bulk region in DC cast ingot.

### Tensile testing near the solidus temperature

Tensile testing was conducted using a Gleeble 3800 thermomechanical testing unit with a low-force load cell at a strain rate of  $10^{-4}$  s $^{-1}$ . Each sample was heated to the desired temperature at a rate of 2°Cs $^{-1}$  and maintained at the testing temperature for 60 s before tensile testing. The temperature evolution during heating and tensile testing was monitored and controlled using a K-type thermocouple spot-welded at the center of each sample. At least two tests were conducted under each condition to confirm the reliability of the results.

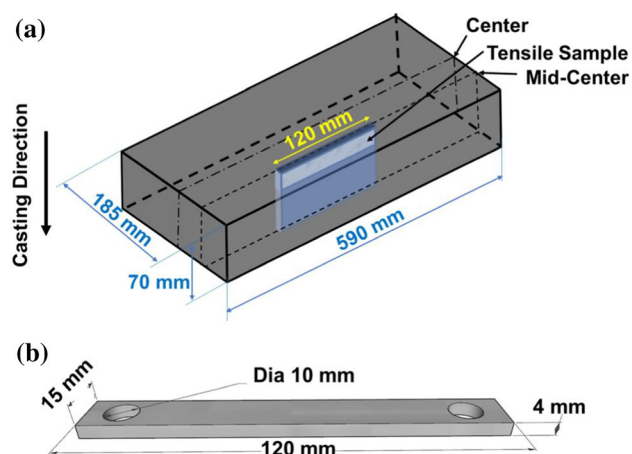
Accurate measurement of the flow stress at high temperatures was challenging because the stress of Al alloys at near-solidus temperatures was low, often ranging between 0.5 and 10 MPa. In this study, a newly developed method was used to calculate the force using the changes in L-gauge displacements [35]. This method allowed us to accurately measure the flow stress in a very narrow range and obtain consistent stress values, particularly in the semisolid state.

Strain measurements during the tensile testing were performed using the DIC method [35]. The displacement of the sample surface was monitored using a monochrome digital camera ( $\alpha$ 7RIII, Sony) mounted on a cannon tripod (Fig. 2a) connected to a remote digital system to control the distance between the camera lens and the sample surface. A speckled pattern was created after spraying quick-dry graphite lubricant (Jig-A-Loo) onto the sample surface. The field of view on the sample surface was approximately 7.2 mm  $\times$  4.8 mm. Images were captured at a rate of 3–4 frames per second and converted to grayscale patterns of 800  $\times$  450 pixels. Thereafter, the images were analyzed using GOM Correlate software (Germany). The parameters used for the analysis

**Table 1** Chemical composition of as-received DC cast ingots

Alloy	Si	Mg	Cu	Fe	Mn	Ti
A	0.7	0.6	0.7	0.2	0.2	0.03
B	0.6	0.6	0.5	0.2	0.1	0.03
Standard AA6111	0.6–1.1	0.5–1	0.5–0.9	0.4 max	0.1–0.45	0.1 max





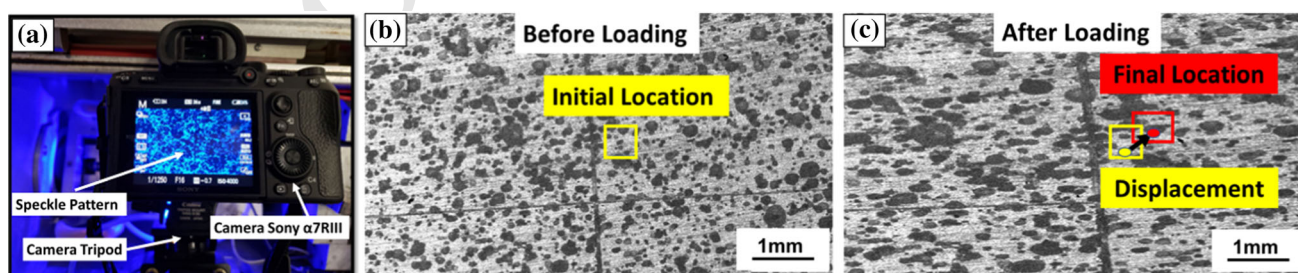
**Figure 1** **a** Positions of metallographic and tensile samples in DC cast ingots and **b** geometry and dimension of tensile test sample.

220 were as follow: subset size of  $33 \times 33$ , step size of 12,  
 221 bicubic subpixel interpolation and resolution of 0.01  
 222 pixel or  $0.09 \mu\text{m}$ . The step size is the distance between  
 223 the center of the subset and the closest neighbor  
 224 subset [36]. It is crucial to increase the spatial reso-  
 225 lution of DIC by decreasing the step size to obtain an  
 226 acceptable strain map for localized strain [37, 38]. The  
 227 subset size was also selected to include at least three  
 228 particles (Fig. 2b, c) [36, 38]. The DIC method  
 229 involves comparing a reference image (before the  
 230 tensile test at zero strain) with the images of the  
 231 deformed samples. The measured strains were the  
 232 averages of three points along the centerlines of the  
 233 tensile samples, where the temperatures were the  
 234 exact test temperatures. The stress–strain curves of the  
 235 tensile samples were created by synchronizing the  
 236 stress measured using the L-gauge method with the  
 237 strain at fracture determined using the DIC method  
 238 based on their evolution over time.

## Metallography analysis

239  
 240 For microstructural characterization, ingot samples  
 241 were subjected to a standard metallographic polish-  
 242 ing procedure [39]. Microstructural examinations  
 243 were performed using an optical microscope and a  
 244 scanning electron microscope (SEM, JSM-6480 LV)  
 245 equipped with energy-dispersive spectroscopy (EDS)  
 246 apparatus. Three samples sliced from the mid-center  
 247 area of the ingots were used to measure the volume  
 248 fractions of the different intermetallics (Fig. 1). The  
 249 minimum cross-sectional area of each metallographic  
 250 specimen was  $160 \text{ mm}^2$ , according to ASTM E45.  
 251 Fifty images per specimen were used in this experi-  
 252 ment. In addition, samples subjected to the tensile  
 253 testing were sliced normal to the loading direction to  
 254 investigate their fracture surfaces. The fractured  
 255 samples were investigated in directions normal and  
 256 parallel to the loading direction to study the fracture  
 257 surfaces and areas surrounding the cracks. The  
 258 examination of the area surrounding the crack (par-  
 259 allel to loading direction) can provide the details  
 260 about the start of liquid film formation and frag-  
 261 mentation of intermetallic particles, while the cross-  
 262 sectional area of fractured tensile samples (normal to  
 263 loading direction) can show the initiation of the main  
 264 crack responsible for the fracture.

265 For porosity measurements, six random samples  
 266 with surface areas of  $160 \text{ mm}^2$  were sliced from the  
 267 mid-center areas of the ingots. One hundred SEM  
 268 images were captured at  $30 \times$  magnification and  
 269 analyzed using the ImageJ software to evaluate their  
 270 porosity percentages. Six samples were sliced from  
 271 the mid-center region of each ingot to study the  
 272 evolution of porosity/voids at semisolid tempera-  
 273 tures. The samples were heated isothermally in the  
 274 temperatures range of  $510\text{--}580 \text{ }^\circ\text{C}$ , followed by water  
 275 quenching. Next, SEM images of the samples were



**Figure 2** **a** Tensile test setup with the camera for 2D digital image correlation, **b** the reference image of the tensile sample surface prior to the test and **c** the deformed image during tensile test.

276 obtained at  $30 \times$  magnification. The specimens were  
 277 sliced 5 mm from the fracture surface parallel to the  
 278 loading direction to measure the porosity percentages of  
 279 the fractured tensile samples.

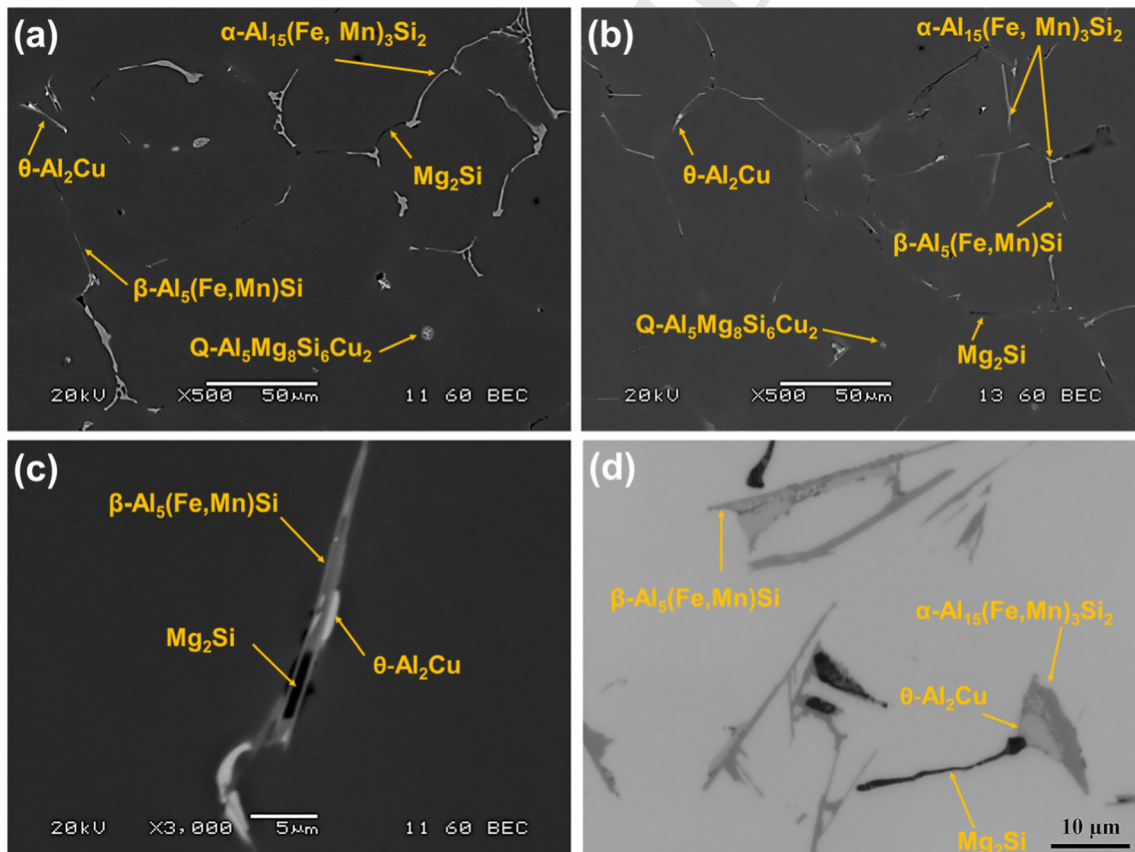
## 280 Results

### 281 Microstructure of the cast ingots

282 The as-cast microstructures of the alloys are shown in  
 283 Fig. 3. The microstructures of both alloys comprised  
 284  $\alpha$ -Al dendrite cells and several intermetallic phases  
 285 concentrated in the interdendritic regions, including  
 286 primary  $Mg_2Si$ , two Fe-rich intermetallics ( $\alpha$ - $Al_{15}$ -  
 287  $Fe, Mn_3Si_2$  and  $\beta$ - $Al_5(Fe, Mn)Si$ ), and two Cu-bearing  
 288 intermetallics ( $Q$ - $Al_5Mg_8Si_6Cu_2$  and  $\theta$ - $Al_2Cu$ ). The  
 289 intermetallic phases were identified based on their  
 290 morphologies and SEM-EDS analysis results. The  
 291 dark lamellar regions in the SEM images (Fig. 3a, b)  
 292 are attributed to the primary  $Mg_2Si$  phase, whereas  
 293 the bright areas are attributed to the Fe-rich and Cu-

bearing phases. The morphologies and  $(Fe + Mn)/Si$  294  
 ratios of the particles were used to identify the Fe-rich 295  
 intermetallic phases. The  $\alpha$ - $Al_{15}(Fe, Mn)_3Si_2$  phase 296  
 exhibited Chinese-script morphologies, and their 297  
 $(Fe + Mn)/Si$  ratios were approximately 1.5. In contrast, 298  
 the  $\beta$ - $Al_5(Fe, Mn)Si$  phase exhibited a platelet- 299  
 like shape, and its  $(Fe + Mn)/Si$  ratio was found to 300  
 be  $\sim 0.8$ . The results of SEM-EDS analysis showing 301  
 the chemical compositions of different phases are 302  
 listed in Table 2. The primary  $Mg_2Si$  and  $Al_2Cu$  303  
 phases nucleated on the surfaces of the Fe-rich 304  
 intermetallics (Fig. 3c) and often grew close to the Fe- 305  
 rich intermetallic phases in the interdendritic region 306  
 (Fig. 3d).

The phase precipitation and temperature during 308  
 solidification predicted by the Scheil model for 309  
 AA6111 alloys were reported to be comparable to the 310  
 experimental values determined using two thermal 311  
 analysis methods [40]. Hence, the Scheil model was 312  
 used to estimate the solidification path and calculate 313  
 the fraction of the solid vs. temperature curves of the 314



**Figure 3** Microstructure of received cast ingots in the mid-center region, **a** Alloy A and **b** Alloy B, **c**  $Mg_2Si$  and  $Al_2Cu$  nucleated on Fe-rich intermetallics, and **d** presence of  $Mg_2Si$  and  $Al_2Cu$  close to Fe-intermetallic in the interdendritic region.

**Table 2** The results of SEM–EDS analysis showing the chemical compositions of different phases (at%)

Phases	Al	Si	Fe	Mg	Mn	Cu	Fe + Mn/Si (at.%)
$\alpha$ -Al <sub>15</sub> Fe,Mn <sub>3</sub> Si <sub>2</sub>	75.9	9.5	9.4	–	5.2	–	1.5
$\beta$ -Al <sub>5</sub> FeSi	80.9	10.6	8.2	–	0.3	–	0.8
Q-Al <sub>5</sub> Mg <sub>8</sub> Si <sub>6</sub> Cu <sub>2</sub>	67.1	12.0	–	14.4	–	6.5	–
$\theta$ -Al <sub>2</sub> Cu	81.8	–	–	–	–	18.2	–
Mg <sub>2</sub> Si	–	30.5	–	69.5	–	–	–

alloys studied, using the Thermo-Calc software with the TCAL7 database. The results are shown in Table 3 and Fig. 4. According to the solidification path predicted by the Scheil model, after the formation of  $\alpha$ -Al dendrites,  $\alpha$ -Al<sub>15</sub>(Fe,Mn)<sub>3</sub>Si<sub>2</sub> precipitated at 614–617 °C, whereas  $\beta$ -Al<sub>5</sub>(Fe,Mn)Si precipitated at 588–594 °C. Furthermore, Mg<sub>2</sub>Si was formed as the binary eutectic of  $\alpha$ -Al + Mg<sub>2</sub>Si at 555–560 °C, as well as the ternary eutectic of  $\alpha$ -Al + Mg<sub>2</sub>Si + Si at 533–536 °C. Two Cu-bearing intermetallic phases precipitated at lower temperatures, near the solidus temperature. The formation temperatures of the Q-Al<sub>5</sub>Mg<sub>8</sub>Si<sub>6</sub>Cu<sub>2</sub> and  $\theta$ -Al<sub>2</sub>Cu intermetallics were 529 and 510 °C, respectively. In addition, some studies reported that the binary eutectic  $\alpha$ -Al +  $\alpha$ -AlFeMnSi solidified in the range of 609–634 °C [40, 41], and binary eutectic ( $\alpha$ -Al + Mg<sub>2</sub>Si) grew preferentially on the surface of  $\beta$ -AlFeMnSi forming a ternary eutectic ( $\alpha$ -Al + Mg<sub>2</sub>Si +  $\beta$ -AlFeMnSi), as shown in Fig. 3c, d [42–44].

The area fractions of the Fe-rich and Cu-bearing intermetallics in the microstructure of the two alloys were different because the Si, Cu, and Mn contents of the alloys were different. The quantitative metallographic analysis results of the alloys are shown in Fig. 5. The area fractions of Chinese-script  $\alpha$ -Al<sub>15</sub>(Fe,Mn)<sub>3</sub>Si<sub>2</sub> of alloy A were significantly higher than

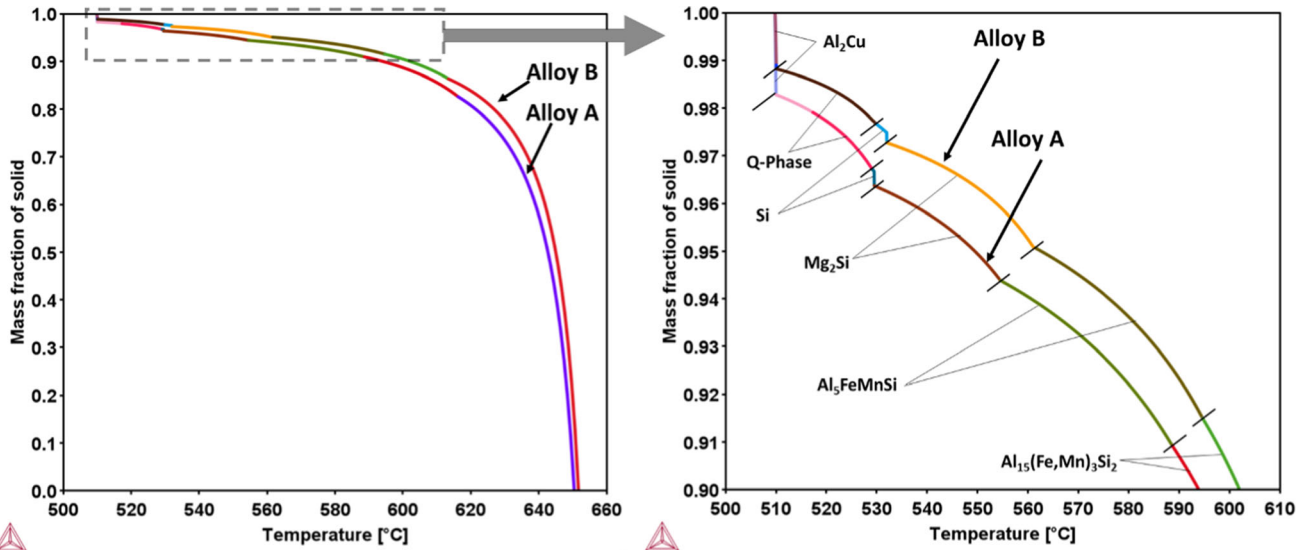
those of alloy B. Adding Mn to Al alloys modifies the morphology of Fe-rich intermetallics from platelets to Chinese-script and increases the volume fraction of Fe-bearing intermetallics [45]. In contrast, platelet-like  $\beta$ -Al<sub>5</sub>(Fe,Mn)Si is the primary Fe-rich bearing intermetallic in alloy B. The area fraction of the Fe-rich intermetallics increased from 1.4% for alloy B to 1.64% for alloy A. Furthermore, the fraction of the Mg<sub>2</sub>Si phase of alloy A was greater than that of alloy B. In addition, owing to the higher Cu content of alloy A, the fraction of Cu-bearing phases in alloy A was higher than that in alloy B. In brief, the area fractions of the low-melting-point eutectic phases (Mg<sub>2</sub>Si, Q, and Al<sub>2</sub>Cu) of alloy A were significantly higher than those of alloy B.

### Porosity/void formation at semisolid temperatures

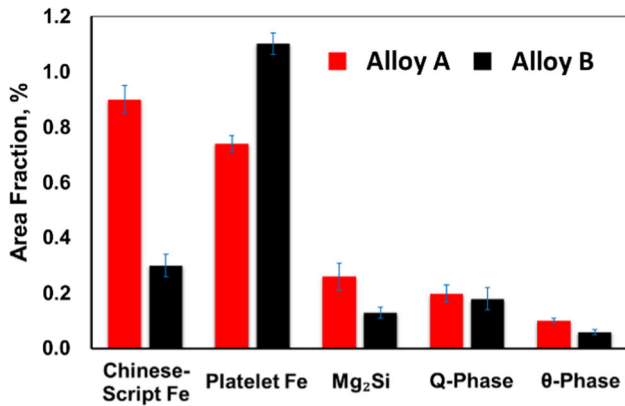
The porosity and void formation were studied at various temperatures in the semisolid temperature range (510–580 °C). The original porosities in the microstructures of the DC cast ingots were similar and very low (< 0.1%, Fig. 6a, b, and e). Upon increasing the temperature from 25 to 510 °C, the porosity percentage of alloy A increased from 0.08% to 0.3%, whereas that of alloy B increased from 0.07 to

**Table 3** Solidification path of two alloys predicted from Scheil model and comparison with results in literature

Solidification Path	Current study		Larouche et al. [41]		Chen et al. [40]	
	Alloy A	Alloy B	DSC	Scheil model	Thermal analysis	DSC
1. $l \rightarrow \alpha - \text{Al} + \alpha - \text{Al}(\text{Fe}, \text{Mn})\text{Si}$	616.5	614.7	621–632	630–634	609–632	633
2. $l + \alpha - \text{Al}(\text{Fe}, \text{Mn})\text{Si} \rightarrow \alpha - \text{Al} + \beta - \text{Al}(\text{Fe}, \text{Mn})\text{Si}$	588.5	594.7				606
3. $l \rightarrow \alpha - \text{Al} + \text{Mg}_2\text{Si}$	555.5	560	545–548	541–547	553–555	557
4. $l \rightarrow \alpha - \text{Al} + \text{Mg}_2\text{Si} + \text{Si}$	533	536				
5. $l \rightarrow \alpha - \text{Al} + \text{Q} - \text{Phase}$	529	529	527–538	537		
6. $l \rightarrow \alpha - \text{Al} + \text{Q} - \text{Phase} + \theta - \text{Al}_2\text{Cu} + \text{Si}$	510	510	507–515	510	506	508



**Figure 4** Evolution of mass solid fraction during solidification of both alloys calculated by the Scheil model.



**Figure 5** Quantitative results of the area fractions of different intermetallic phases in two alloys.

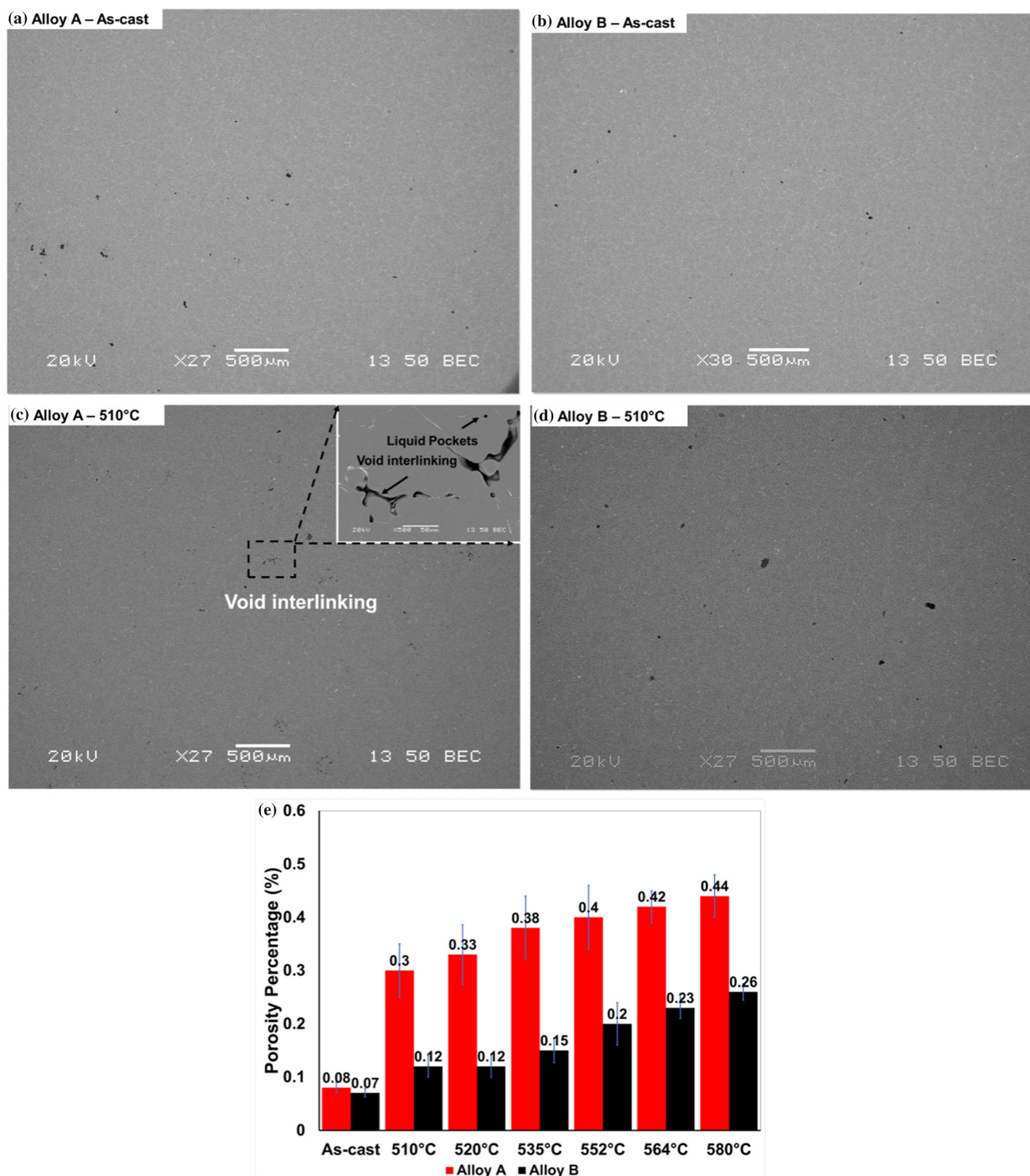
367 0.12% (Fig. 6c, d, and e). The increase in porosity  
 368 percentage and void interlinking degree were more  
 369 significant for alloy A than for alloy B. The increase of  
 370 the 0.05% porosity content in alloy B corresponded  
 371 well with the content of Al<sub>2</sub>Cu (i.e., 0.06%) of the  
 372 sample. However, for the alloy A, the porosity was  
 373 increased by (0.22%), which was higher than the  
 374 Al<sub>2</sub>Cu content of the sample. The significant increase  
 375 in porosity content in alloy A implies that the  
 376 Q-phase started to melt at such temperature, and  
 377 therefore, the void interlinking is more significant in  
 378 alloy A (Fig. 6c). Upon further increasing the tem-  
 379 perature to the upper semisolid range (580 °C), the  
 380 highest temperature for the tensile tests, the porosity  
 381 percentage increased gradually with temperature for  
 382 both alloys. However, the porosity of alloy A was

383 significantly higher than that of alloy B over the  
 384 entire temperature range, attributed to higher  
 385 amounts of low-melting-point eutectic phases (e.g.,  
 386 Mg<sub>2</sub>Si, Q, and Al<sub>2</sub>Cu) in the as-cast microstructure of  
 387 alloy A than in alloy B. In addition, at semisolid  
 388 temperatures, the porosity of alloy A became irregu-  
 389 lar along the dendrite boundaries and shrinkage-  
 390 like pores formed (Fig. 6c). In contrast, the pores of  
 391 alloy B increased in size without forming shrinkage-  
 392 like pores (Fig. 6d). The high number of pores and  
 393 changes in the porosity morphology of alloy A at  
 394 high semisolid temperatures, in the absence of  
 395 external tensile forces, suggest that alloy A is highly  
 396 sensitive to hot-tearing evolution.

### Mechanical properties at near-solidus temperature

#### Engineering stress–strain curves

397  
 398  
 399  
 400 The typical engineering stress–strain curves of alloys  
 401 A and B were obtained in the temperature range of  
 402 450–580 °C and a strain rate of  $\sim 10^{-4} \text{ s}^{-1}$  (Fig. 7). The  
 403 results of the test conducted at 450 °C were used to  
 404 represent the solid-state tensile flow behavior of the  
 405 alloys. In general, the flow stress increased sharply  
 406 toward the peak stress. After reaching the peak  
 407 stress, the flow stress progressively decreased to the  
 408 fracture point (Fig. 7a). The solid-state strength and  
 409 ductility of alloy A were higher and lower, respec-  
 410 tively, than those of alloy B, attributed to the content



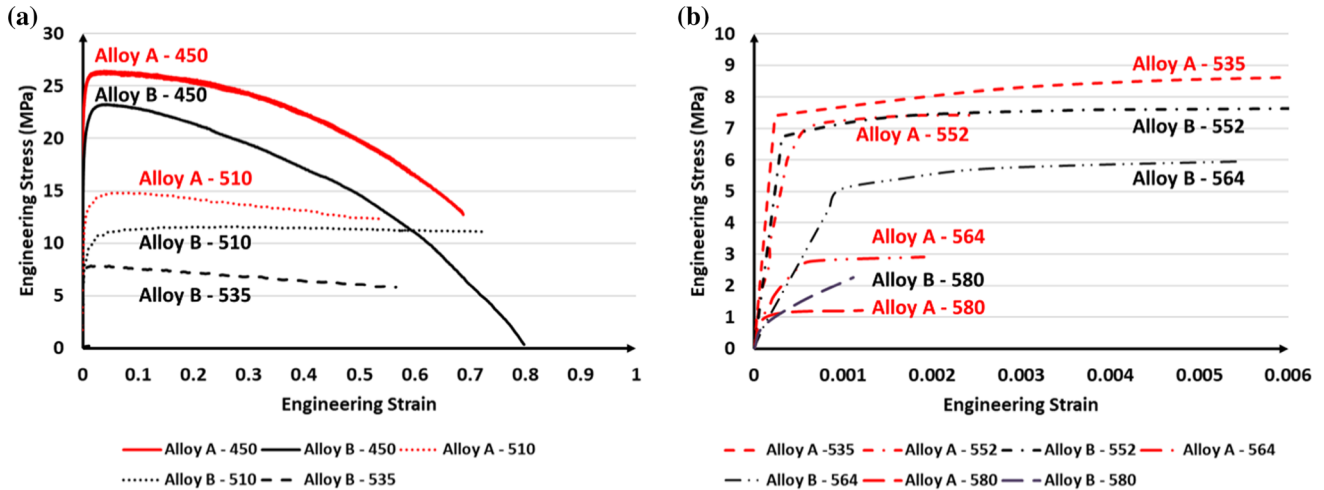
**Figure 6** a As-cast porosity in alloy A, b as-cast porosity in alloy B, c porosity and voids in alloy A at 510 °C with a enlarged view in the inset, d porosity and voids in alloy B at 510 °C and e evolution of porosity with increasing temperature in both alloys.

411 of intermetallic phases, comprising Fe-rich and Cu-  
 412 bearing intermetallics. The  $Mg_2Si$  phase of alloy A  
 413 was higher than that of alloy B. Upon increasing the  
 414 temperature to 510 °C (near the solidus temperature),

both alloys exhibited plateaus in stress after reaching  
 the peak stress, continuing until the fracture point  
 was reached. The ductility of alloy A remained lower  
 than that of alloy B.

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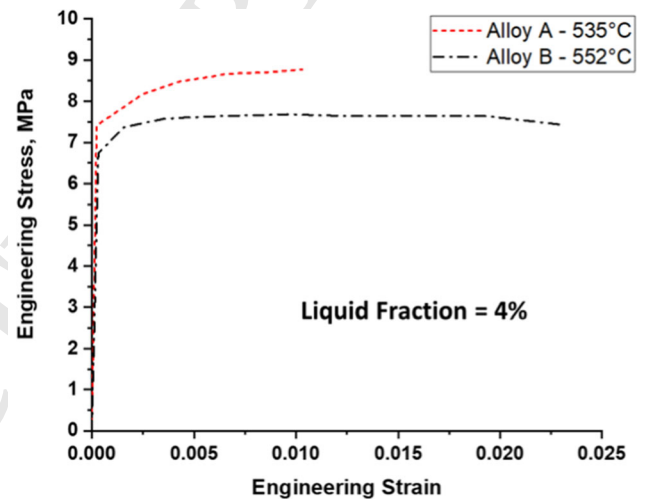




**Figure 7** Engineering stress–strain curves for two AA6111 alloys **a** at 450–535 °C and **b** the enlarged view at the low strain for the temperatures between 535 and 580 °C.

419 Upon increasing the temperature to 535 °C, the  
 420 ductility (strain at failure) of alloy A decreased  
 421 sharply (3.2%), whereas alloy B exhibited a signifi-  
 422 cantly greater strain at failure (approximately 60%)  
 423 than alloy A (Fig. 7b). Moreover, the tensile strengths  
 424 of the alloys were similar at 552 °C; however, the  
 425 ductility of alloy B (2.8%) was considerably higher  
 426 than that of alloy A (0.25%). All tensile properties are  
 427 summarized in Table 4.

428 For a given mass fraction of liquid, for instance at  
 429 an  $f_L$  of 4%, the strength of the mush structure of  
 430 alloy A reached its maximum value much faster than  
 431 that of alloy B; however, the strain at failure of alloy  
 432 A was lower than that of alloy B (Fig. 8). The area  
 433 under the engineering stress–strain curve of a mate-  
 434 rial is equivalent to the modulus of toughness, rep-  
 435 resenting the strain energy per unit volume required  
 436 to fracture the material [46]. The higher the absorbed  
 437 energy required for crack growth, the larger the  
 438 stored strain energy [47, 48]. Accordingly, at  $f_L = 4%$ ,  
 439 the resistance of alloy B to crack propagation was



**Figure 8** Engineering stress–strain curves for two AA6111 alloys at a constant liquid fraction of 4%.

440 significantly higher than that of alloy A (the strain  
 441 energy of alloy B ( $0.172 \text{ MJm}^{-3}$ ) was higher than that  
 442 of alloy A ( $0.086 \text{ MJm}^{-3}$ )).

**Table 4** Tensile properties of alloys A and B over the whole temperature range

Temperature, °C	Ultimate tensile strength, MPa		Failure strain, %	
	Alloy A	Alloy B	Alloy A	Alloy B
450	26.30 ± 0.80	23.30 ± 0.90	69.30 ± 5.20	80.30 ± 4.20
510	20.40 ± 0.90	16.40 ± 0.80	61.10 ± 4.12	76.20 ± 5.10
535	15.60 ± 0.50	12.50 ± 0.90	3.20 ± 0.20	67.80 ± 3.62
552	7.40 ± 0.30	7.70 ± 0.70	2.50 ± 0.15	2.80 ± 0.20
564	2.90 ± 0.40	5.90 ± 0.60	1.80 ± 0.02	0.50 ± 0.02
580	1.20 ± 0.30	2.30 ± 0.20	0.10 ± 0.01	0.10 ± 0.01

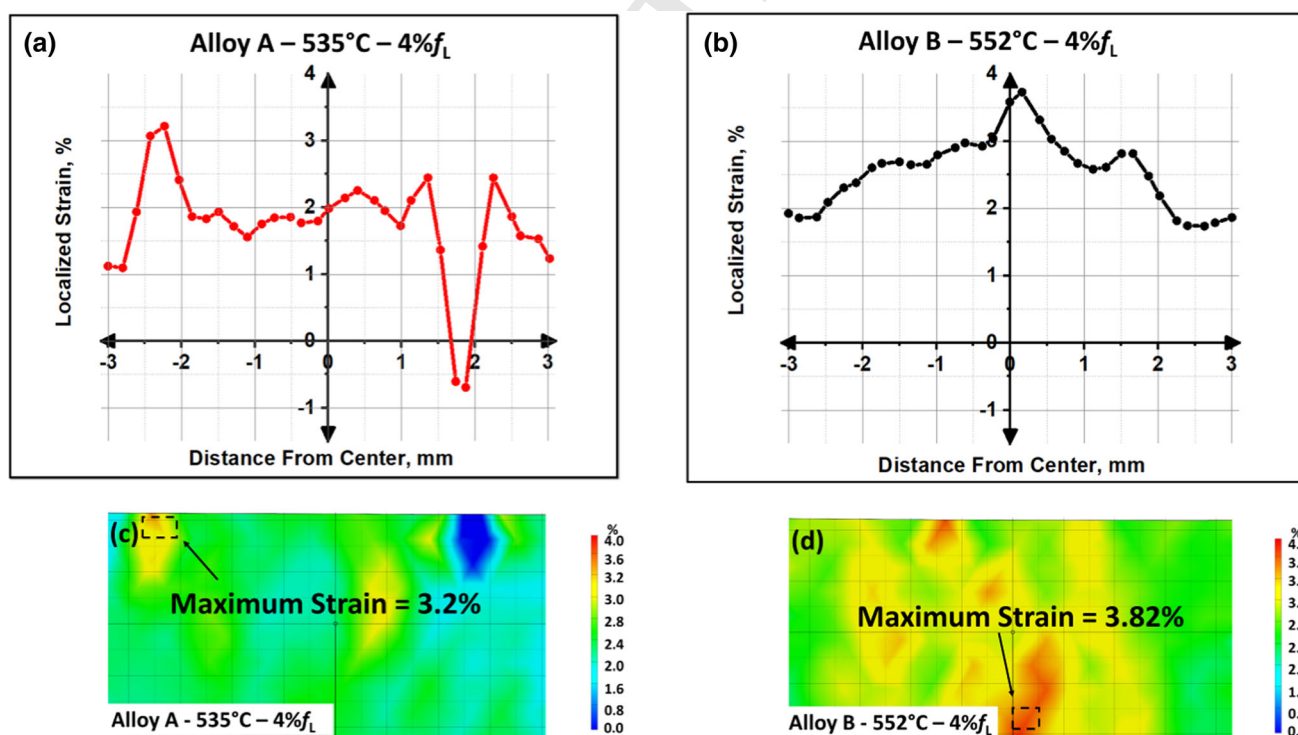
443 *Strain field at fracture zone*

444 The strain maps of alloys A and B with the same  $f_L$  of 4% are presented in Fig. 9. The strain distribution along the sample length indicated that the location of the maximum strain for alloy A was random, and the peak strain appeared away from the sample center ( $\sim 2$  mm). Crack opening is expected to occur in the hotspot zone, where the strain should be maximum [35]. In other words, cracking should be located at the centerline where the temperature is the highest. However, for crack-susceptible alloys, the hot tear can be initiated at the weakest points of the sample (e.g., porosity and voids). Therefore, the initiation outside the hotspot zone for alloy A was attributed to pre-existing defects promoting strain accumulation [49]. The maximum strains at fracture for alloys A and B are 3.2% and 3.8%, respectively. As the  $f_L$  values of both alloys were the same, alloy A, which fractured at a lower strain, was more sensitive to pre-existing defects than alloy B.

463 Figure 10 shows the time evolution of the strain for the two alloys at two  $f_L$  values. As shown, the strain rates of the alloys were considerably different. For a  $f_L$  of 4%, after 15 s, the strain of alloy A reached a

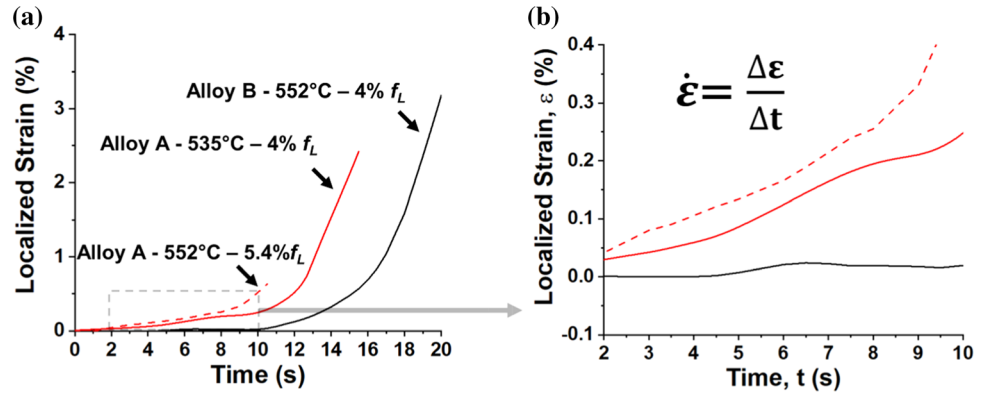
467 maximum of 3.2%, whereas that of alloy B reached only 0.5% (compared with its total strain of 3.82%). The strain rate of alloy A was considerably much higher than that of alloy B. After 11 s of tensile testing, the strain of alloy B approached 0%, whereas that of alloy A was approximately 0.35%. In addition, alloy A exhibited a steep increase in strain near the fracture point, whereas alloy B exhibited a gradual increase in strain until the fracture point. The higher strain rate of alloy A was attributed to its greater sensitivity to hot tearing.

478 Figure 10a shows the strain rate of alloy A remained unchanged with increasing temperature from 535 to 552 °C, whereas its strain at failure decreased significantly. It was also found that alloy B exhibited higher failure strain compared with alloy A at a constant mass fraction of liquid (i.e., 4% $f_L$ ). The enlarged view of the strain evolution at the early stage of the curves (i.e., time  $\leq 10$  S) is presented in Fig. 10b. At 552 °C, the strains ( $\epsilon$ ) and strain rates ( $\dot{\epsilon}$ ) of the alloys at the same strength were significantly different; moreover, alloy A fractured before strain localization began in alloy B, indicating that the cracks were propagated in alloy A even prior to initiation of the cracking in alloy B.



**Figure 9** Strain fields and maps at a constant liquid fraction (4%), a, c Alloy A, b, d Alloy B, a, b strain distribution along sample lengthwise direction before fracture, and c, d strain map showing strain contours before fracture.

**Figure 10** **a** The strain evolution for alloys A (dashed line) and B (solid line) with time, and **b** enlarged view of the rectangular area indicated in Fig.a.

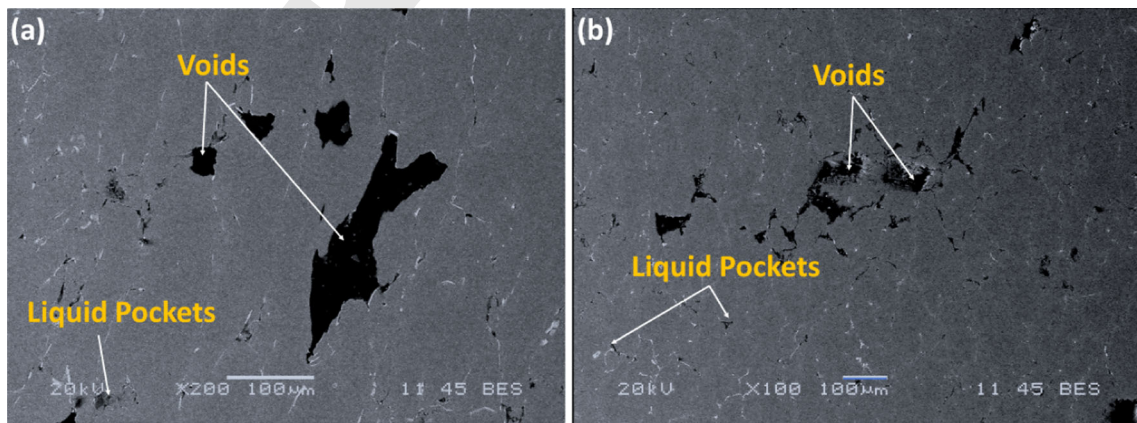


492 **Fracture surface analysis**

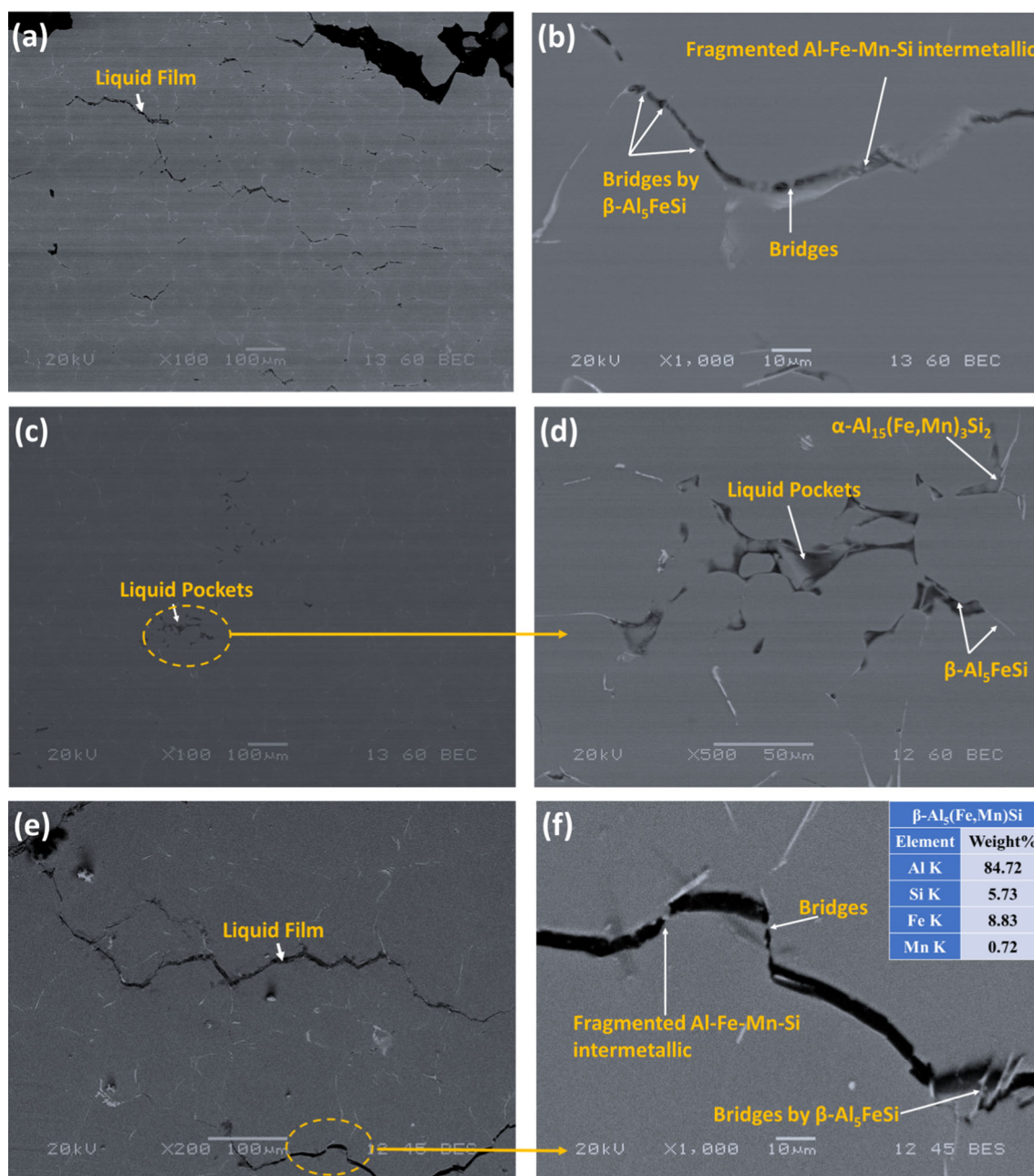
493 Figure 11 shows the area normal to the fracture surfaces of the tensile samples tested at 510 °C (the  
 494 solidus temperature). The liquid pockets formed in  
 495 the microstructure at 510 °C (Fig. 6) were trans-  
 496 formed into large voids, indicating that the fracture  
 497 mechanism was based on void coalescence. As  
 498 mentioned in Sect. 3.2, shrinkage porosity and voids  
 499 were readily formed in alloy A at this temperature.  
 500 By measuring the porosity percentage of the frac-  
 501 tured tensile samples of both alloys, the porosity  
 502 percentage of alloy A (0.65%) was higher than that of  
 503 alloy B (0.5%). Void coalescence was also more sig-  
 504 nificant in alloy A, with the maximum void size of  
 505 280  $\mu\text{m}$ ; in contrast, the maximum void size of alloy B  
 506 was only 80  $\mu\text{m}$ . As mentioned in Sect. 3.2, the sus-  
 507 ceptibility of alloy A to the porosity formation during  
 508 isothermal holding was higher than that of alloy B.  
 509 Therefore, during heating to 510 °C, followed by the  
 510 subsequent isothermal holding before tensile testing,  
 511

shrinkage pores formed in alloy A; therefore, voids  
 grew faster in alloy A than in alloy B. AQ5 513

Figure 12 shows the starting points of liquid film  
 formation via coalescence the pre-existent liquid  
 pockets for both alloys. By investigating the fractured  
 tensile sample of alloy A at 535 °C, a liquid film with  
 a significant width started to develop (Fig. 12a, b).  
 Conversely, the liquid pockets in alloy B started to  
 coalesce, forming intergranular cracks or shrinkage  
 porosity at 535 °C (Fig. 12c and d). No traces of liquid  
 film were present in the cracks, suggesting that the  
 intergranular liquid films were very thin. Moreover,  
 liquid pockets were distributed along the Fe-rich  
 intermetallics. Upon further increasing the tempera-  
 ture from 535 to 552 °C, the liquid films were dis-  
 tributed along the grain boundaries of alloy B  
 (Fig. 12e, f). The width of the liquid film of alloy B at  
 552 °C (4.95  $\mu\text{m}$ ) was approximately two times larger  
 than that of alloy A at 535 °C (2.15  $\mu\text{m}$ ). As the width  
 of the liquid film for alloy A was smaller, more  
 bridges readily formed across the cracks by the Fe-  
 rich intermetallics and  $\text{Mg}_2\text{Si}$  phase (Fig. 12b). Large 512  
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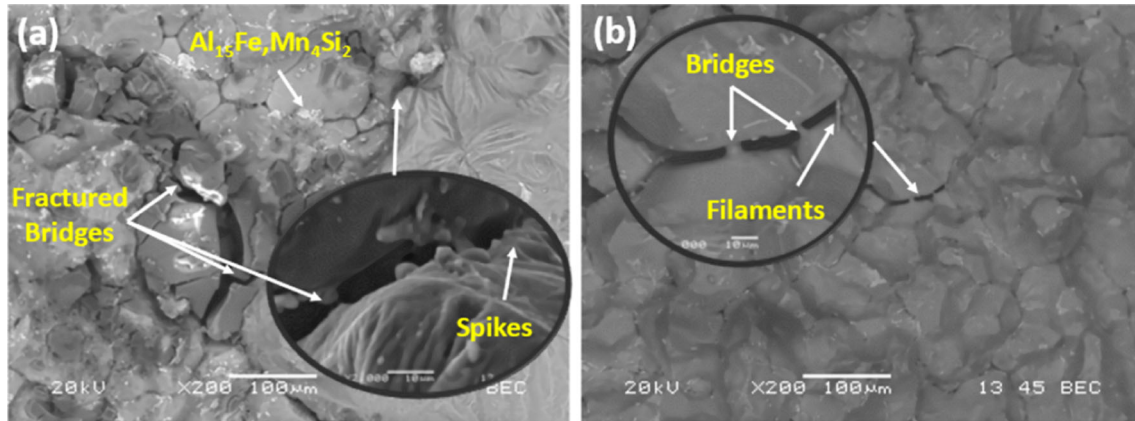
**Figure 11** Appearance of liquid pockets in samples tested at 510 °C: **a** alloy A and **b** alloy B.



**Figure 12** Appearance of liquid film along grain boundaries at 4%  $f_L$  for **a, b** alloy A tested at 535 °C, **c, d** the coalescence of liquid pockets forming cracks in alloy B at 535 °C, and **e, f** alloy B tested at 552 °C.

534 platelet-like Fe-rich intermetallics formed bridges in  
 535 the alloy B, whereas small fragmental Fe-rich inter-  
 536 metallics formed bridges in the alloy A (Fig. 12b and  
 537 f). The small width of the liquid films of alloy A  
 538 facilitated the formation of solid bridges by the small  
 539 Fe-rich intermetallics. It was expected that the small  
 540 fragments of Fe-intermetallics would stop forming  
 541 bridges upon a further increase in liquid film width  
 542 with increasing temperature.

The fracture surfaces of alloys A and B at 580 °C 543  
 are presented in Fig. 13. At this temperature, both the 544  
 alloys exhibited low strength and ductility. The 545  
 fractured bridges observed on the fracture surface of 546  
 alloy A—marked by arrows in Fig. 13a—indicate that 547  
 the intermetallics could not sustain the strength of the 548  
 mush structure at temperatures in this range. Fur- 549  
 thermore, the Fe-rich intermetallics were mostly 550  
 fragmented and could not sustain the strength of the 551



**Figure 13** Fracture surface of samples tested at 580 °C for **a** Alloy A and **b** Alloy B.

552 specimens. In addition, the liquid film thickening of  
 553 alloy A was more severe than that of alloy B because  
 554 the strength of alloy A decreased considerably faster  
 555 (Fig. 7b). The width of the liquid film of alloy A  
 556 increased to 6.6  $\mu\text{m}$ , with Fe-rich intermetallics no  
 557 longer able to form intact solid bridges. The spikes  
 558 observed around the fractured bridges are attributed  
 559 to the extremely localized ductility of alloy A.

560 Nevertheless, alloy B retained its strength, even at  
 561 580 °C. The width of the liquid film of alloy B did not  
 562 change significantly at 580 °C (5.3  $\mu\text{m}$ ). The small  
 563 width of the liquid film facilitated bridge formation,  
 564 and the presence of an unbroken platelet-like  $\beta$ -  
 565 AlFeMnSi phase (Fig. 13b) increased the sample  
 566 strength and delayed crack propagation.

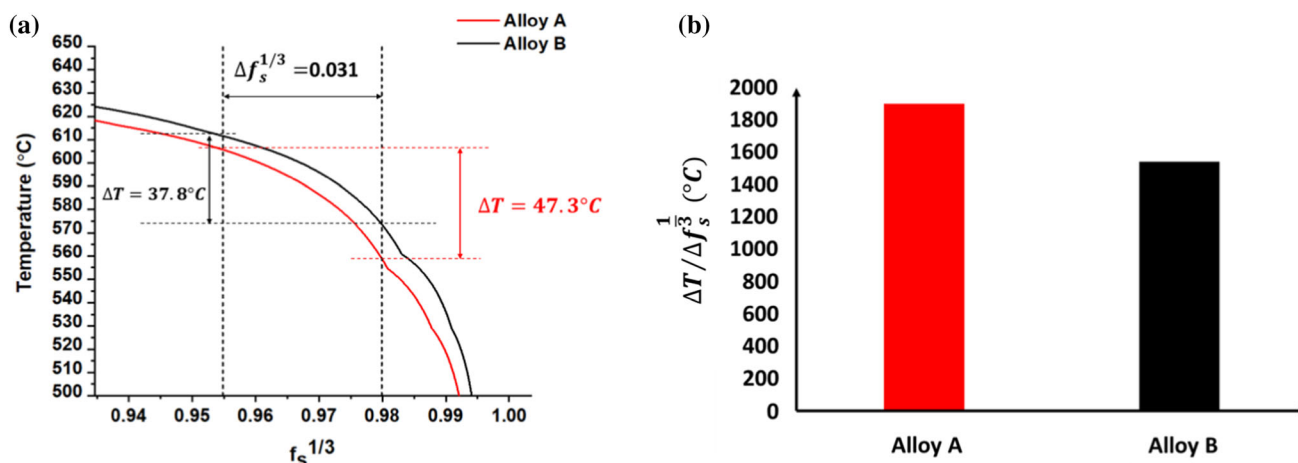
## 567 Discussion

568 The microstructures and mechanical responses of  
 569 AA6111 DC cast alloys (A and B) in the semisolid  
 570 state at high  $f_s$  values were significantly different.  
 571 Two criteria were used to study the HTS behavior of  
 572 the alloys. The first, based on a non-mechanical cri-  
 573 terion proposed by Kou [25, 31] and modified by Hu  
 574 et al. [50], was introduced to establish a relationship  
 575 between HTS and the chemical compositions of the  
 576 alloys. This criterion is based on the evolution of the  
 577 solid fraction with temperature during the late stage  
 578 of solidification. The second criterion is based on a  
 579 mechanical model used to identify the BTR of the  
 580 alloys [10, 32, 34]. The semisolid behaviors and HTS  
 581 indices of the alloys were analyzed using these  
 582 **AQ6** approaches.

## Non-mechanical criterion

583 Kou used the the crack sensitivity factor ( $dT/df_s^{1/2}$   
 584 value near  $(f_s)^{1/2} = 1$ ) to evaluate the HTS behavior of  
 585 alloys with columnar grain structures [25, 51]. Hu  
 586 et al. [50] subsequently modified the model, as  $dT/$   
 587  $df_s^{1/3}$  near  $(f_s)^{1/3} = 1$ , to predict the HTS behaviors of  
 588 alloys with equiaxed grain morphology. Hu's model  
 589 was used because the grain structures of alloys A and  
 590 B were equiaxed. According to Kou, the predicted  
 591 results for peak crack susceptibility ( $\Lambda$ -shaped curve)  
 592 in the solidification range of  $0.87 < f_s < 0.94$  were  
 593 consistent with the experimental results [25, 31]. For  
 594 example, in Al-Si alloys, peak crack susceptibility  
 595 occurred at a Si content of 1%, consistent with the  
 596 hot-tearing data of Al-Si alloys reported by Singer  
 597 [17] and Vero [52]. Therefore, this solidification range  
 598 was adopted in this study. The T vs.  $(f_s)^{1/3}$  curves for  
 599 alloys A and B are shown in Fig. 14a. The calculated  
 600 hot-tearing index ( $\Delta T/\Delta f_s^{1/3}$ ) of alloys A and B were  
 601 1900 and 1540 °C, respectively (Fig. 14b). The pre-  
 602 **AQ7**dicted higher HTS of alloy A is partially attributed to  
 603 its higher Cu content. Furthermore, the HTS indices  
 604 of Al-Mg-Si-Cu alloys increase with increasing Cu  
 605 content [53, 54].

607 The primary drawback of these HTS models is the  
 608 lack of a theoretical basis for selecting a specific  $f_s$   
 609 range; hence, the accuracy of the predicted results  
 610 depends significantly on the selected  $f_s$  range [25, 50].  
 611 In addition, back-diffusion considerably affects the  
 612 high-crack-susceptibility region [31]. As back diffu-  
 613 sion was not considered in these models, the HTS  
 614 behavior of the alloys was further investigated using  
 615 the mechanical criterion.

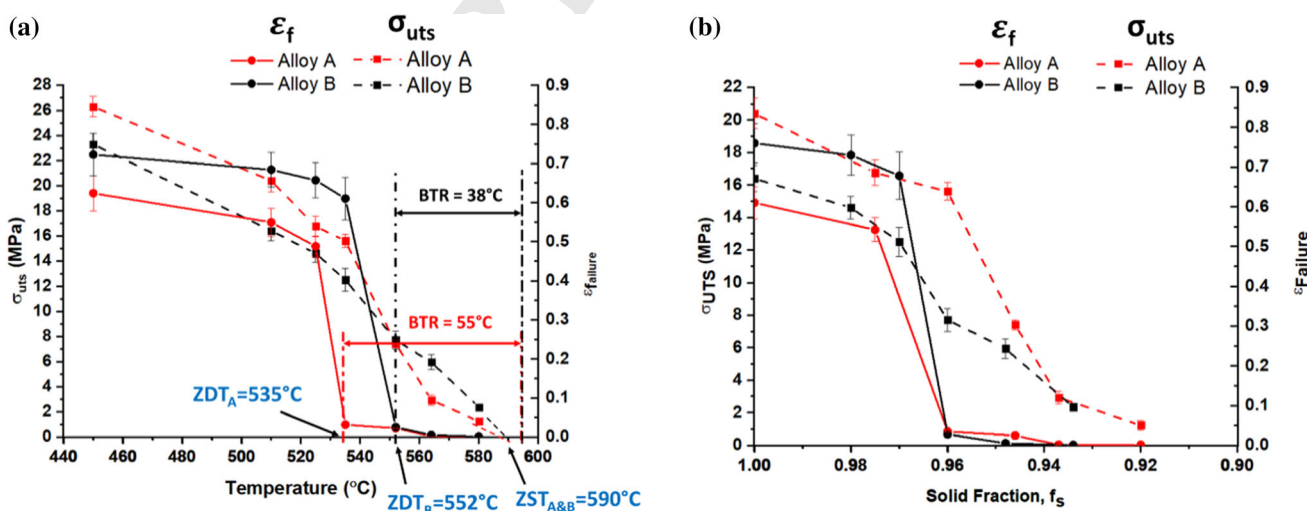


**Figure 14** Prediction of HTS of two AA6111 alloys based on steepness of T versus  $(f_s)^{1/3}$  curves: **a** T versus  $(f_s)^{1/3}$  curves in the  $f_s$  range of 0.87–0.94 ( $0.955 < (f_s)^{1/3} < 0.98$ ); **b** the difference of hot tearing index between alloys A and B.

616 **Mechanical criterion**

617 The relationship between the ultimate tensile strength ( $\sigma_{uts}$ ) and true failure strain ( $\epsilon_f$ ) of the alloys  
 618 with temperature and  $f_s$  is presented in Fig. 15. Three  
 619 zones were observed in the failure strain behavior of  
 620 the alloys over the studied temperature. The first  
 621 zone ranges from the completely solid-state to 525 °C  
 622 (Fig. 15a). Throughout this zone,  $\sigma_{uts}$  and  $\epsilon_f$  decreased  
 623 gradually, and the alloys exhibited a ductile behavior.  
 624 Upon further increasing the temperature, a  
 625 transition zone emerged, where the alloy ductility  
 626 decreased abruptly. At the end of this zone, the  
 627 ductility reached a very low value, and the corre-  
 628 sponding temperature is known as ZDT. It has been  
 629

proposed that ZDT corresponds to the temperature at  
 which the strain is lower than 3% [55]. Therefore, the  
 ZDTs of alloys A and B were determined to be 535  
 and 552 °C, respectively (Fig. 15b). The ZDTs of both  
 the alloys corresponded to the same  $f_s$  value of 0.96.  
 ZDTs of AA6111 alloys have been reported to occur  
 at  $f_s = 0.95$  [6, 7]. However, Phillion et al. [10]  
 demonstrated that the ZDT highly depended on the  
 solidification sequence and the corresponding alloy  
 composition, reporting that the ZDTs of AA6111  
 alloys occurred at  $f_s = 0.99$ . The third zone, known as  
 the BTR, is located above the ZDT and extends to a  
 temperature at which the strengths of the alloys  
 approach zero, known as the ZST. The ZSTs of alloys  
 A and B were estimated to be ~ 590 °C (Fig. 15a),  
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**Figure 15** The UTS ( $\sigma_{uts}$ ) and failure strain ( $\epsilon_f$ ) **a** as a function of temperature and **b** as a function of solid fraction for two semisolid AA6111 alloys.

645 and the corresponding  $f_s$  values of alloys A and B, at  
646 which the ZST was reached, were estimated to be at  
647 approximately 0.92 and 0.91, respectively (Fig. 15b).

648 Throughout the first zone (gradual decrease in  
649 ductility) and above the solidus temperature, liquid  
650 pockets began to form in the alloys due to the melting  
651 of the Cu-bearing low-melting-point eutectic phases.  
652 Moreover, individual liquid pockets were randomly  
653 distributed in the microstructures of the alloys  
654 (Fig. 11). In the second zone, the liquid pockets began  
655 to interlink and coalesce, causing a significant  
656 decrease in ductility. The ZDT was reached at the end  
657 of this zone as continuous liquid films started to form  
658 along the grain boundaries (Fig. 12e and f).

659 As previously mentioned, the BTR zone ranged  
660 from ZDT to ZST; ZST corresponds to the inter-  
661 granular separation stage. Phillion et al. [10] deter-  
662 mined that the ZST of an AA6111 alloy pulled at a  
663 strain rate of  $10^{-4} \text{ s}^{-1}$  was 580 °C, comparable to that  
664 estimated in this study (590 °C). Accordingly, the  
665 BTRs of alloys A and B were calculated to be 55 and  
666 38 °C, respectively (Fig. 15a). Throughout the BTR,  
667 the widths of the liquid films along the grain  
668 boundaries, separation of the bridges across cracks,  
669 and fragmentation of Fe-intermetallics increased  
670 (Fig. 13). The formation of continuous liquid films  
671 significantly decreased the interfacial energy at the  
672 solid-liquid interface, facilitating crack propagation  
673 and hot tearing [21]; therefore, the larger the BTR, the  
674 larger is the HTS index [10]. The strength of alloy A  
675 decreased faster than that of alloy B throughout the  
676 BTR zone, as indicated by the dashed vertical lines in  
677 Fig. 15a. Furthermore, the strength of alloy A was  
678 low ( $< 1 \text{ MPa}$ ) at 580 °C, whereas that of alloy B was  
679 relatively high ( $> 2 \text{ MPa}$ ). The solid bridges in alloy  
680 A were broken at 580 °C; however, the solid bridges  
681 formed by Fe-rich AlFeMnSi intermetallics persisted  
682 in alloy B (Fig. 13).

## 683 Conclusions

684 The semisolid tensile properties at high solid frac-  
685 tions of two AA6111 DC cast alloys with different  
686 chemical compositions were investigated. The Cu,  
687 Mn, and Si contents of alloy A were higher than those  
688 of alloy B. Based on microstructure and semisolid  
689 tensile results, the non-mechanical and mechanical  
690 criteria were used to investigate the HTS behaviors of  
691 the alloys. The results indicated that the HTS index of

alloy A was higher than that of alloy B. The following  
conclusions were drawn:

- 694 1. The incipient melting of the Cu-bearing and  
695  $\text{Mg}_2\text{Si}$  intermetallics during isothermal heating  
696 near and above the solidus temperature (i.e., in  
697 the range of 510–580 °C) caused a sharp increase  
698 in porosity/void formation, thus promoting void  
699 growth during tensile testing in the semisolid  
700 state. The fracture mechanism of the tensile  
701 samples involved void coalescence, which devel-  
702 oped along various intermetallics. As the amount  
703 of low-melting-point eutectic phases (e.g.,  $\text{Mg}_2\text{Si}$ ,  
704 Q, and  $\text{Al}_2\text{Cu}$ ) in the as-cast microstructure of  
705 alloy A was higher than that of alloy B, the  
706 enhanced porosity formation and void interlink-  
707 ing in alloy A were more significant than those in  
708 alloy B.
- 709 2. The tensile strength of both alloys decreased  
710 gradually with increase in temperature, reaching  
711 similar values at 552 °C. At temperatures above  
712 552 °C, the decrease in strength of alloy A was  
713 more significant than that of alloy B. The  $\varepsilon_f$  values  
714 of the alloys decreased sharply with increasing  
715 temperature until ZDT was reached; the ZDT of  
716 alloy A (535 °C) was lower than that of alloy B  
717 (552 °C). At temperatures lower than the ZDT,  
718 the  $\varepsilon_f$  values of alloy B were higher than those of  
719 alloy A.
- 720 3. According to the non-mechanical criterion, the  
721  $dT/df_s^{1/3}$  values of alloy A in the  $f_s$  range of  
722 0.87–0.94 were higher than those of alloy B;  
723 therefore, the HTS index of alloy A was higher  
724 than that of alloy B.
- 725 4. According to the mechanical criterion and using  
726 the ZDT and ZST concepts, the BTR values of  
727 alloys A and B were calculated to be 55 and 38 °C,  
728 respectively. In addition, alloy A exhibited a  
729 sharper decrease in strength in the BTR zone than  
730 alloy B. The wider BTR (45%) and lower strength  
731 of alloy A, associated with significant liquid film  
732 thickening and fragmentation of Fe-rich inter-  
733 metallics, indicated that its HTS was higher than  
734 that of alloy B.

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